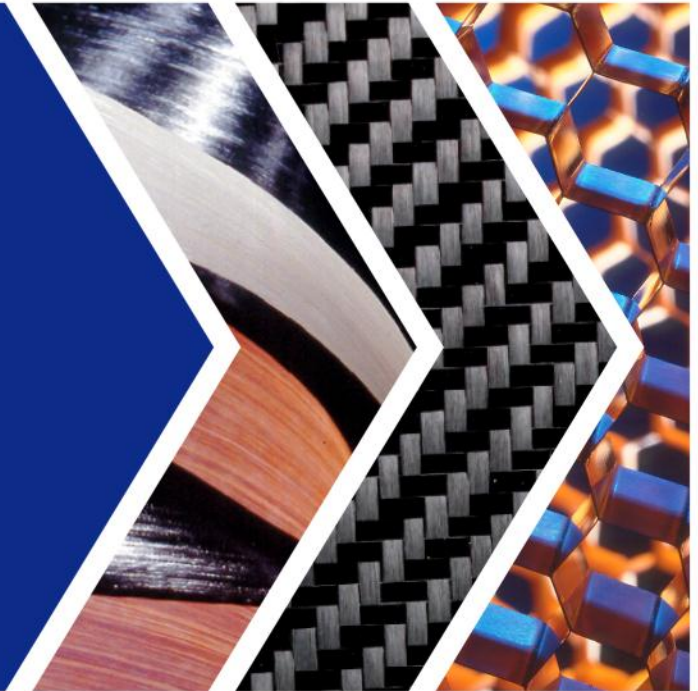




Characterization of Prepreg Materials for Wind Turbine Blades

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9 May 2013



Agenda

- **Hexcel company profile**
- **Background to prepregs in wind energy**
- **Prepreg characterisation**
 - Surface prepregs (XF2P)
 - Structural prepregs (matrix system, M79)
- **Combinations of prepregs with infusion: co-infusion**
- **Conclusions**

Company Profile

- Technology leader in advanced composites
- Serving commercial aerospace, space & defense and industrial
- Net Sales 2012: \$1.58 Billion
- 5,000 employees worldwide
- 19 manufacturing sites (including JV in Malaysia)
- Headquarters in Stamford, CT, USA
- Listed on New York and Paris Stock Exchanges





Background

Prepregs in Wind Energy



Typical Prepreg Systems in Wind Energy

Typical resin systems

M9G 310 J/g

M9GF 230 J/g

M19G 160 J/g

Cure temperature ~100-120°C

UD Products

Carbon 500-600 g/m²

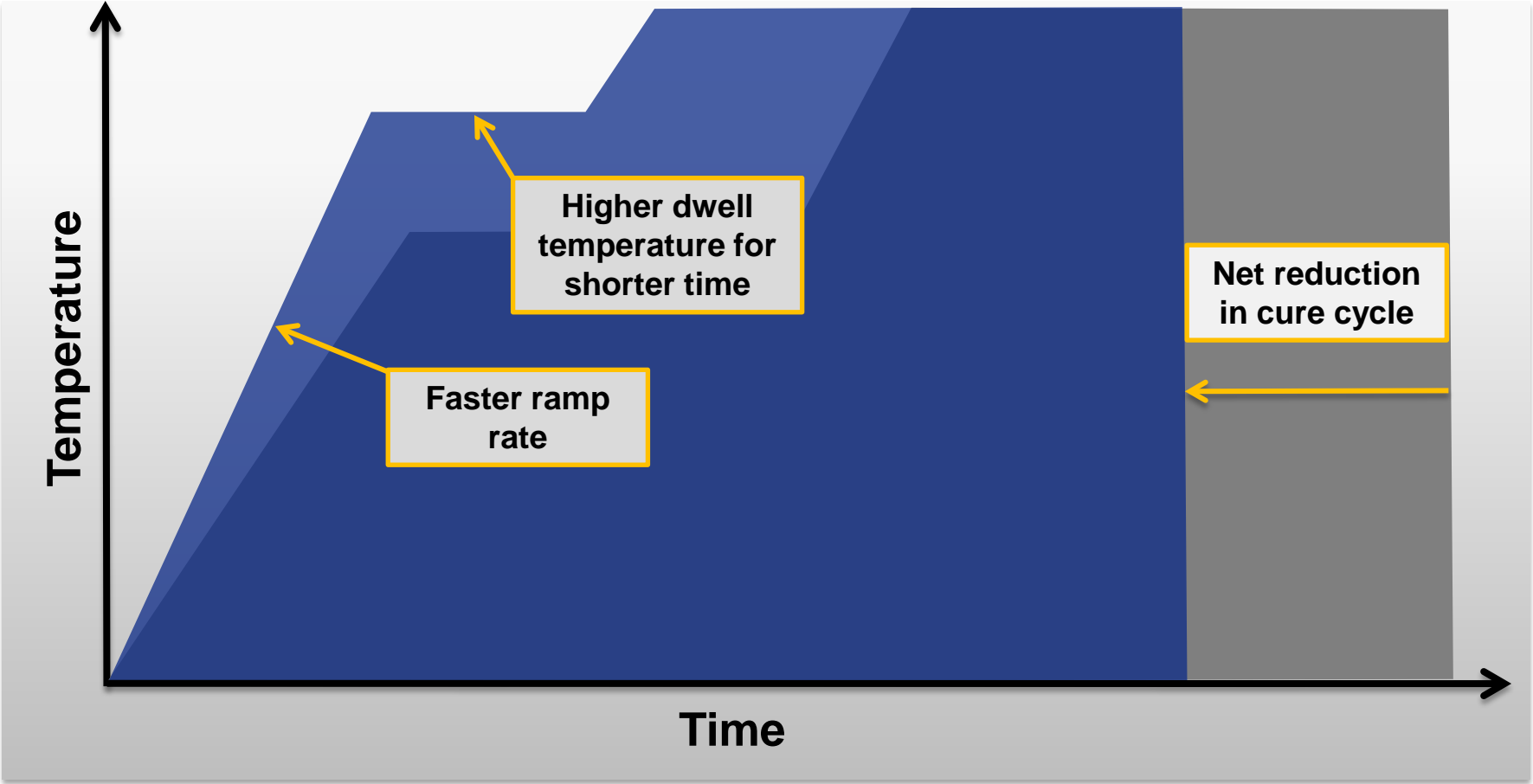
Glass 1000-3000 g/m²

Overall cure cycles

~4 to ~8 hours (optimisation is key)

Typical prepregs
high areal weight + moderate cure temperature + low
reaction enthalpy

The Value of Low Exotherm in Thick Laminates

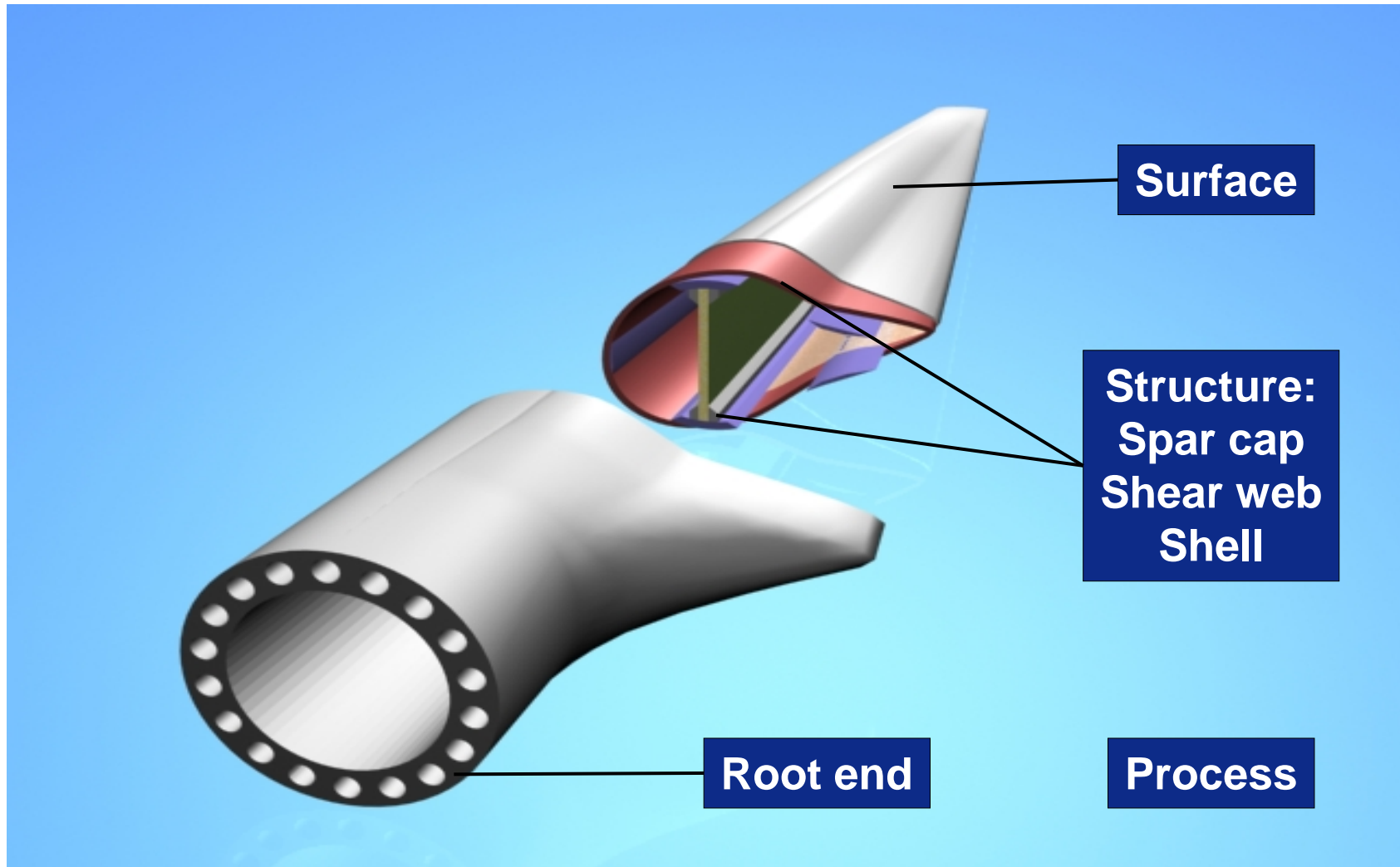


Low exotherm matrix e.g. M19G



Standard exotherm matrix e.g. M9G

Features of Typical Wind Turbine Blades



Prepreg Characterisation

This presentation will focus on the following

Surface and shell prepreg

- **Characterisation of a surface prepreg that obviates an additional gel coat**

Structural prepreg for large/ thick sections

- **Characterisation of a new structural prepreg system, M79, that combines:**
 - Low temperature cure
 - Low exotherm



Characterisation of a Shell Prepreg: HexPly XF2P



Prepregs for the Shell Surface

Shell prepregs are used for the aerodynamic shell

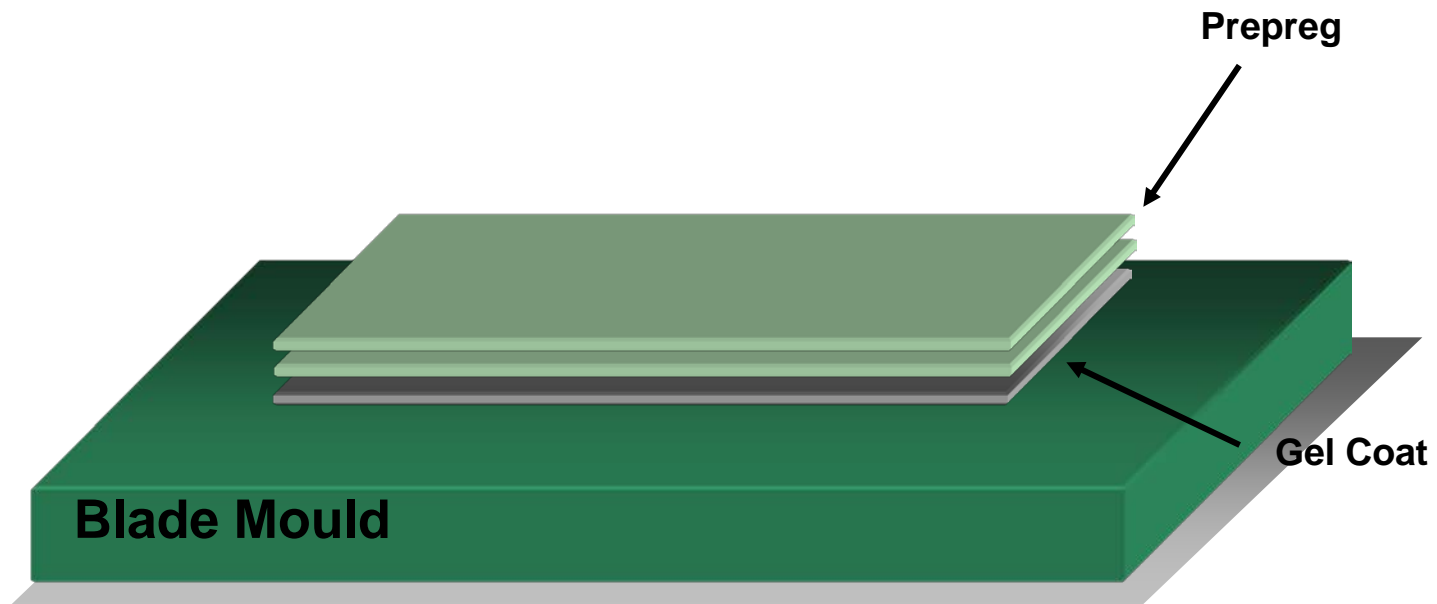
- Gel coats may be used to provide a good paint-ready surface
- Polyurethane paints may be used for the final surface

Painting makes the gel coat redundant as a surface finish system

This process can be simplified by using specific shell prepregs such as HexPly XF2P

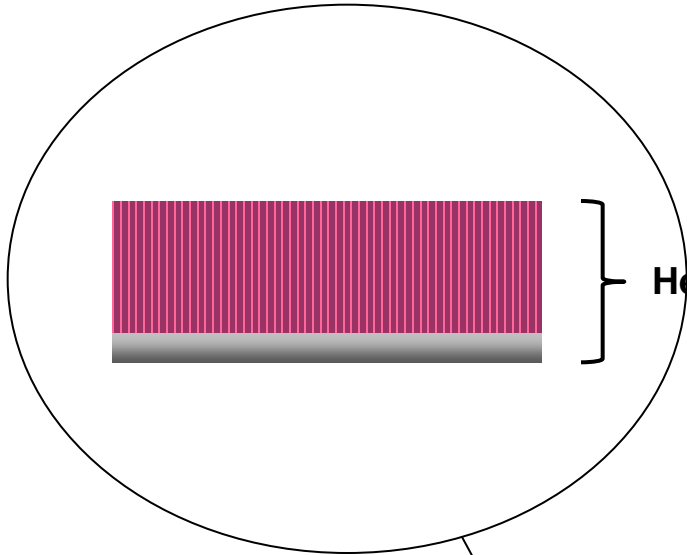
- To build the aerodynamic shell surface
- To eliminate the gel coat

Conventional Shell Construction, with Gel Coat

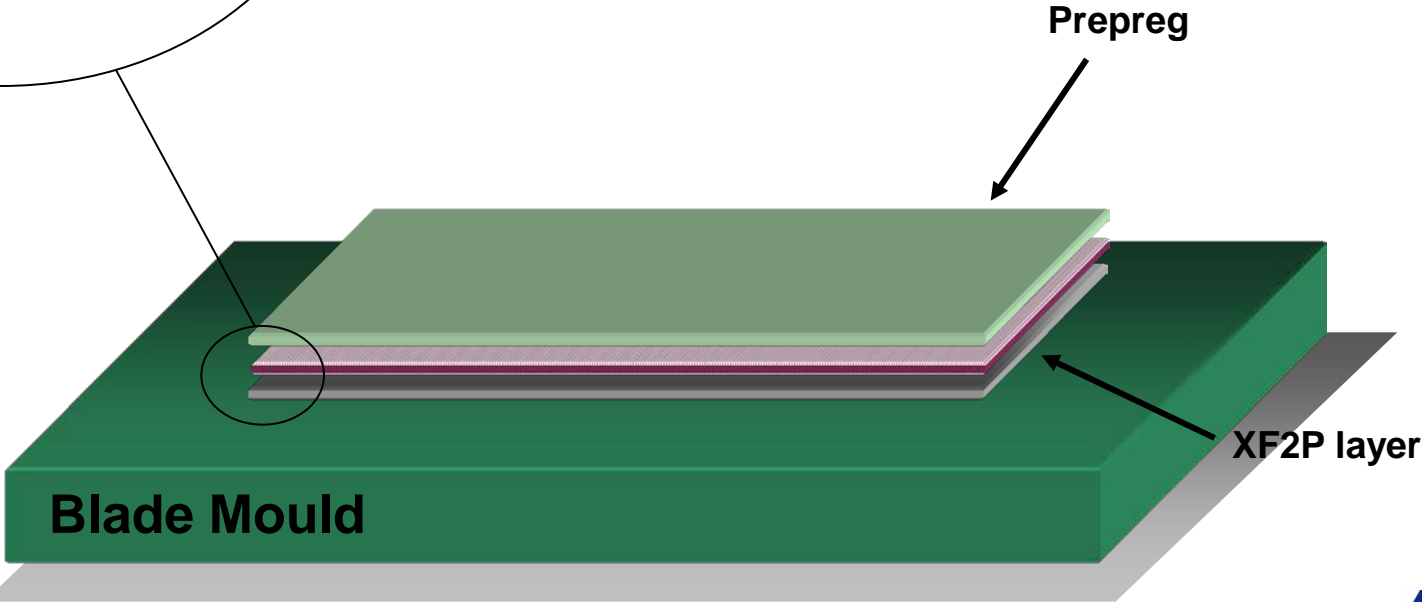


XF2P – Gel Coat-free Surface Finish

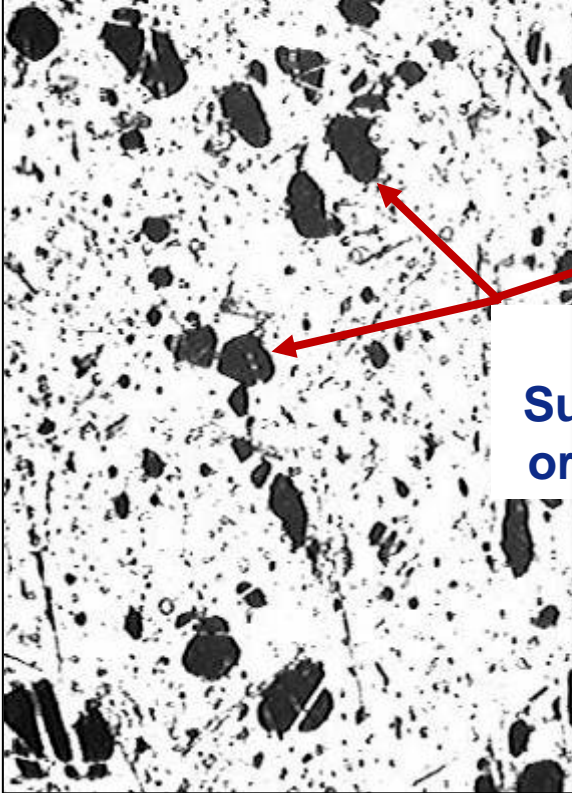
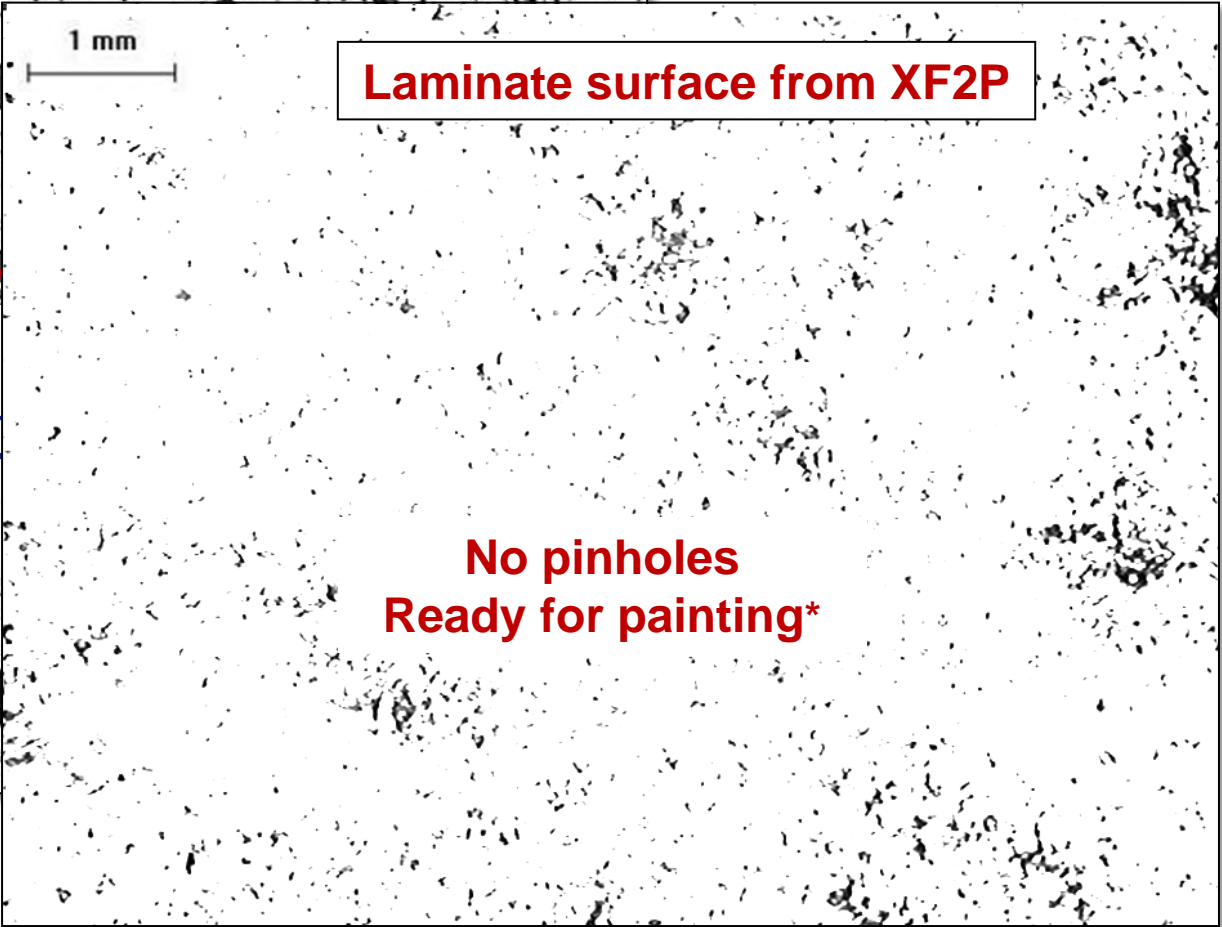
No gel coat needed



HexPly M9.6/ 43%/ LBB1200 + CV/ G

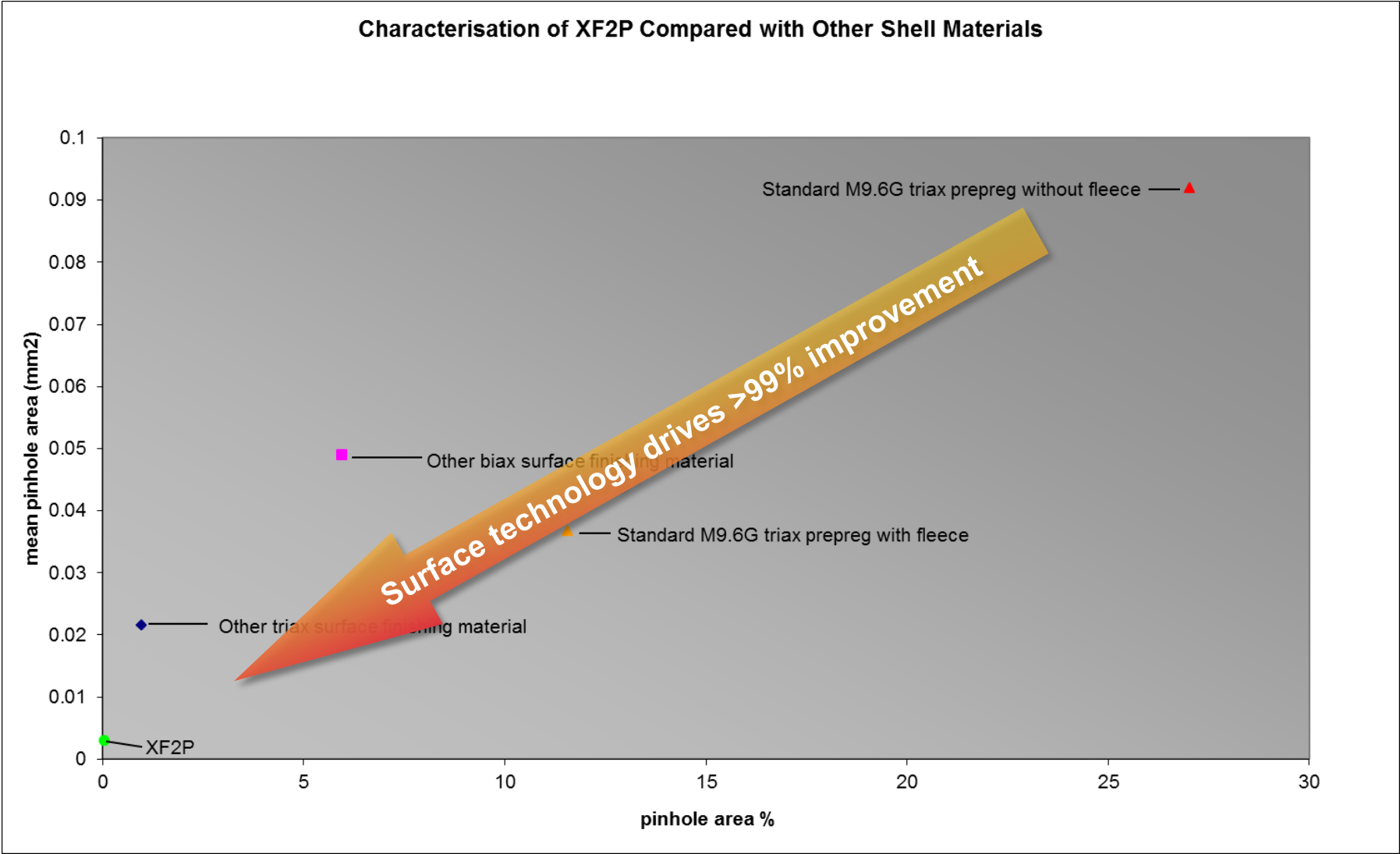


XF2P: Surface Characterisation

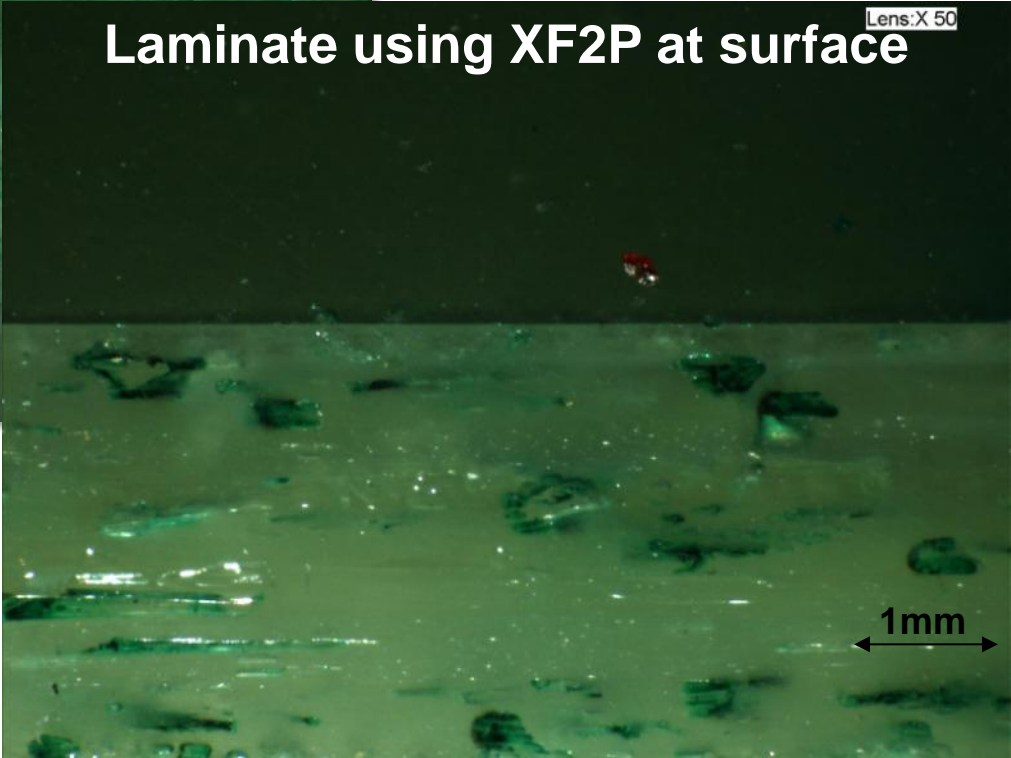
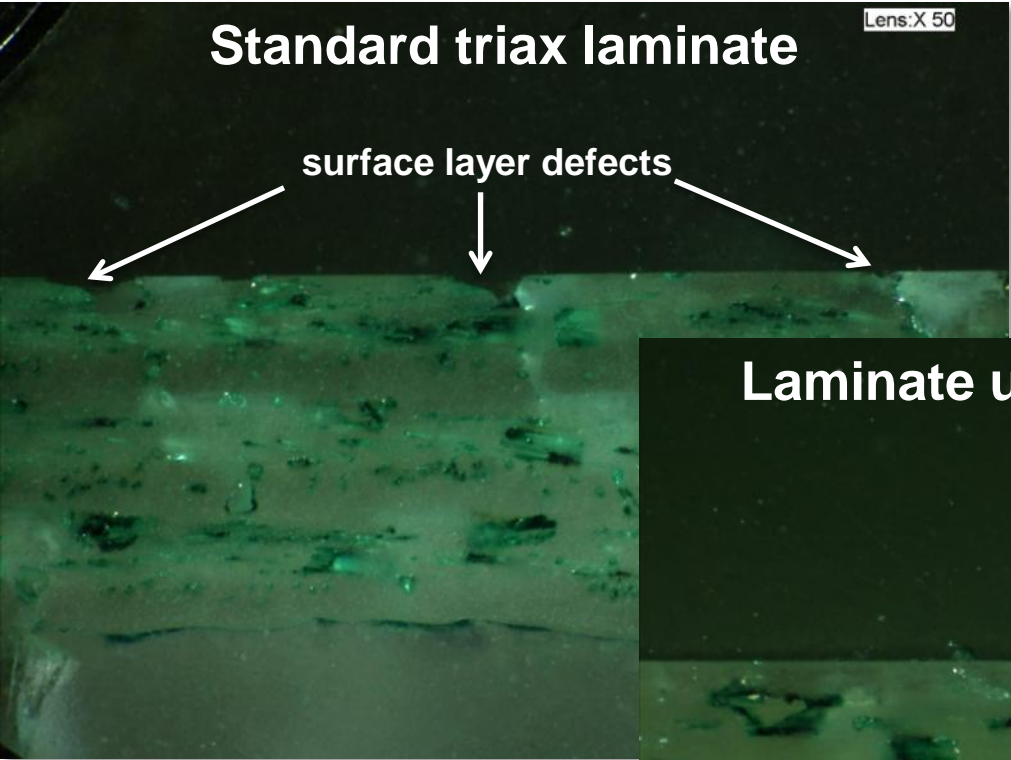


* After removal of release agent

Surface Porosity from Shell Materials



XF2P: Cross-sectional Analysis





Characterisation of Prepreg Matrices: M79

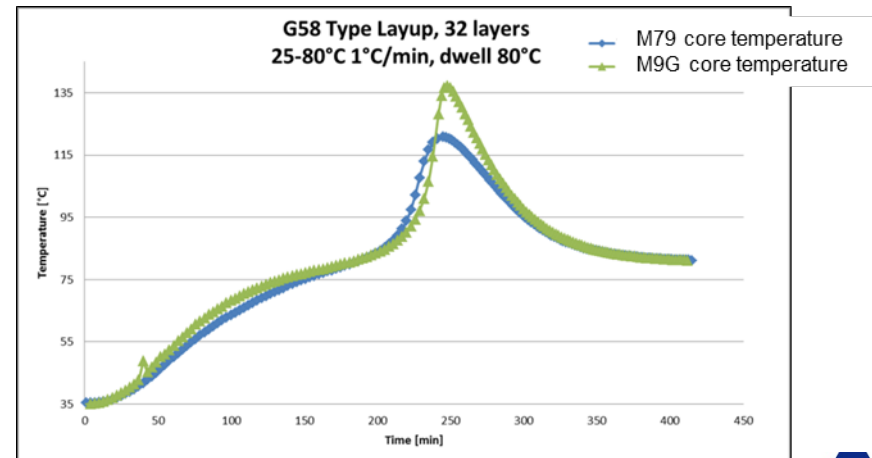
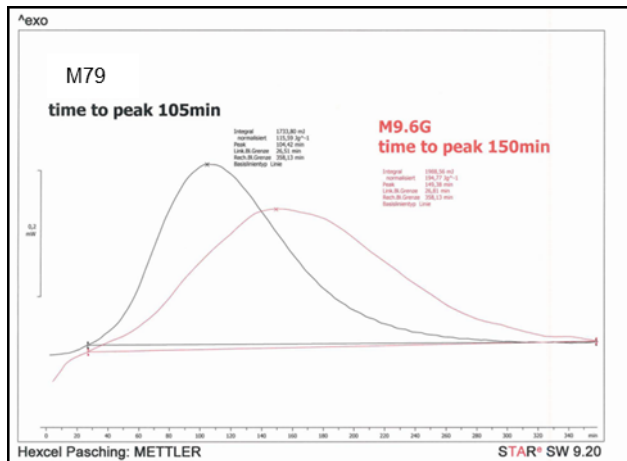
Designed for Structural Applications



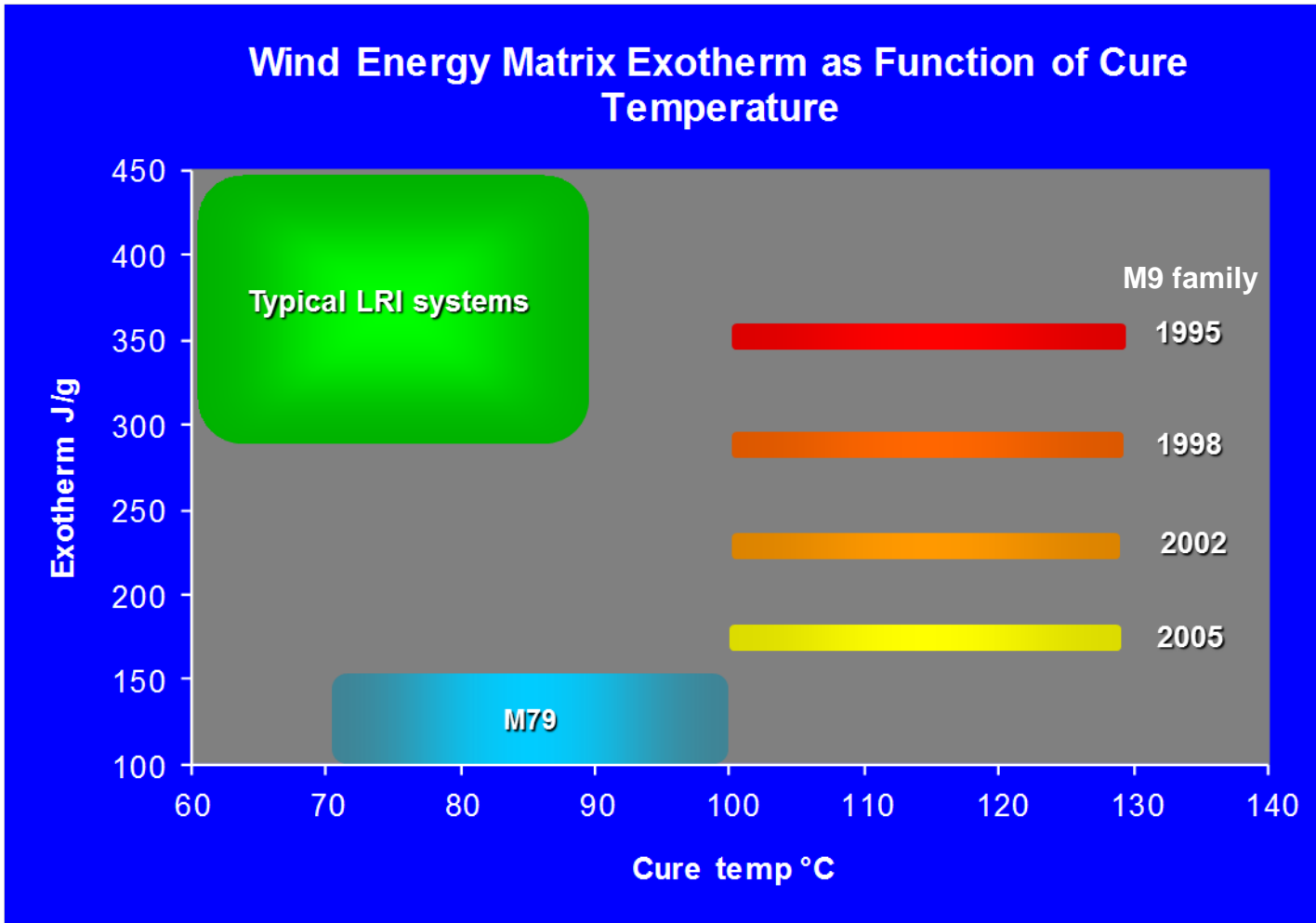
M79

New generation prepreg system for large industrial structures (e.g. wind turbine blades)

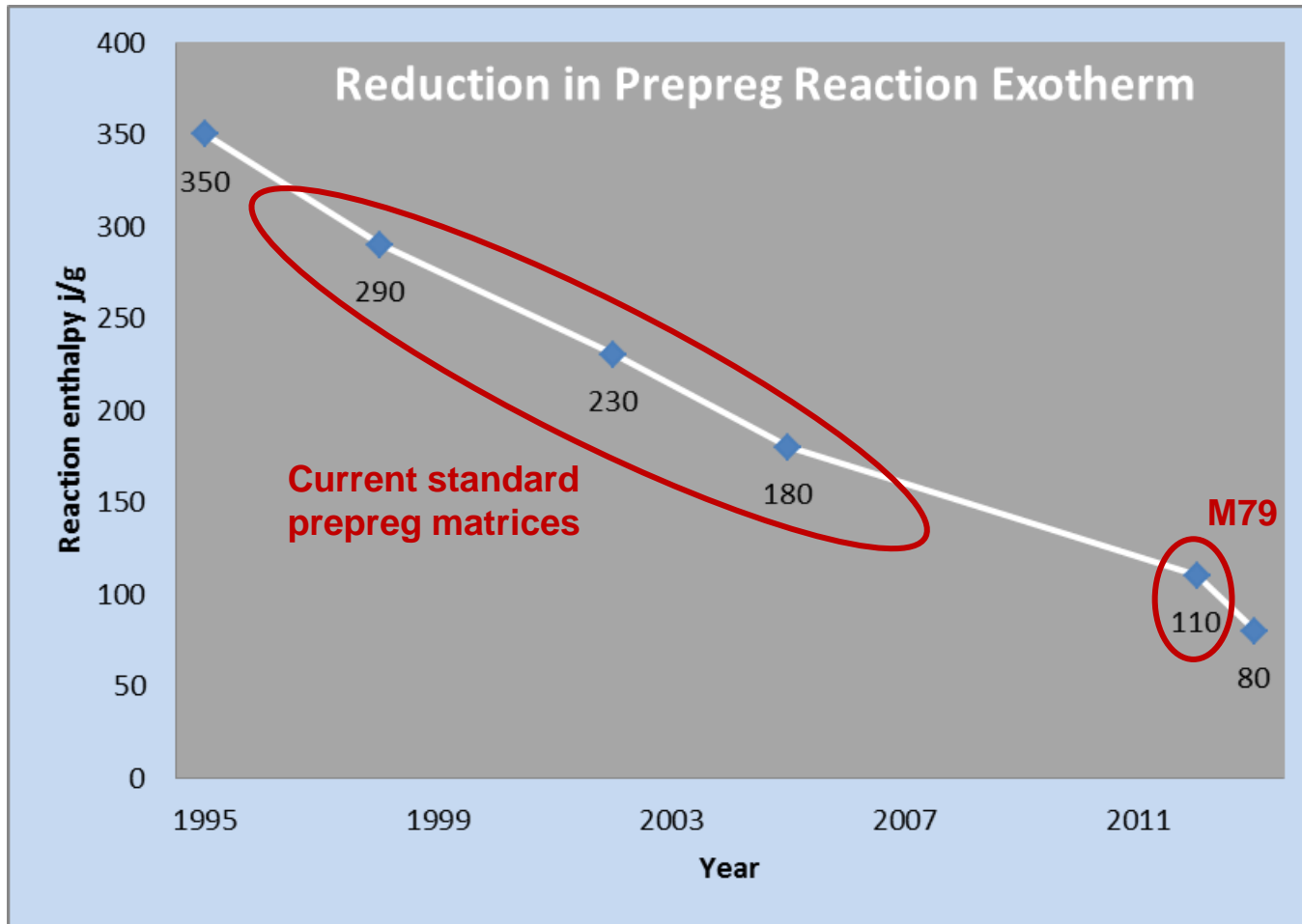
- Cure: 70°C ~10 hrs; 80°C ~6 hrs; 120°C <1 hr
- Outlife > 2 months
- Exotherm ~100-120 j/g
- Static mechanical properties as current M9G family prepregs
- Product form as current prepregs/ semiprepregs
- Manufacture: standard process, as current M9G family prepregs



M79 Compared with Conventional Systems



Reduction in Prepreg Exotherm, 1995-2013



Latest prepreg matrices minimise reaction exotherm allowing short cure cycles of thick structures

M79: Example of Mechanical Test Data

Test & Direction	Measurement	70 ° C Cure			M9 Historical	
		No. of specimens	Mean	SD		CV (%)
Tensile 0°	Strength (MPa)	8	469	9.4	2.0	445
	Modulus (GPa)		21.2	0.5	2.5	18.2
Compression 0°	Strength (MPa)	10	413	20	4.9	333
	Modulus (GPa)		21.0	0.3	1.4	19.5
ILSS (45° , 4-ply)	Strength (MPa)	20	46.7	1.9	4.0	43.6

Normalized results are in bold

Test results for HexPly M79/43%/LBB1200+CV/G cured at 70 °C

Overall, M79 mechanical test data compares favourably with conventional (M9) systems

M79: Example of Mechanical Test Data

Test & Direction	Measurement	80 ° C Cure				M9 Historical
		No. of specimens	Mean	SD	CV (%)	
Tensile 0°	Strength (MPa)	20	456	16	3.6	445
	Modulus (GPa)		19.1	0.3	1.7	18.2
Compression 0°	Strength (MPa)	10	394	30	7.5	333
	Modulus (GPa)		20.5	1.0	4.7	19.5
ILSS (45° , 4-ply)	Strength (MPa)	20	39.5	1.1	2.7	43.6

Normalized results are in bold

Test results for HexPly M79/43%/LBB1200+CV/G cured at 80 °C

Overall, M79 mechanical test data compares favourably with conventional (M9) systems

M79 and Co-infusion

Co-infusion

The use of prepreg and infusion technologies in the same laminate with co-cure

Typical configuration

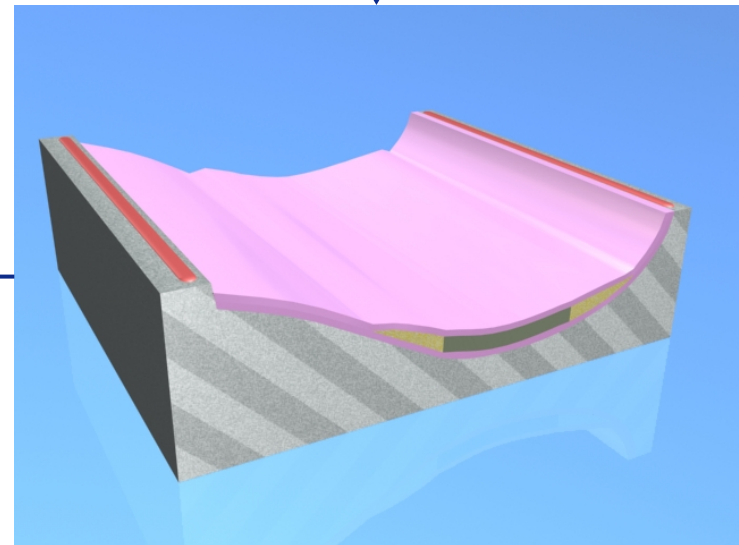
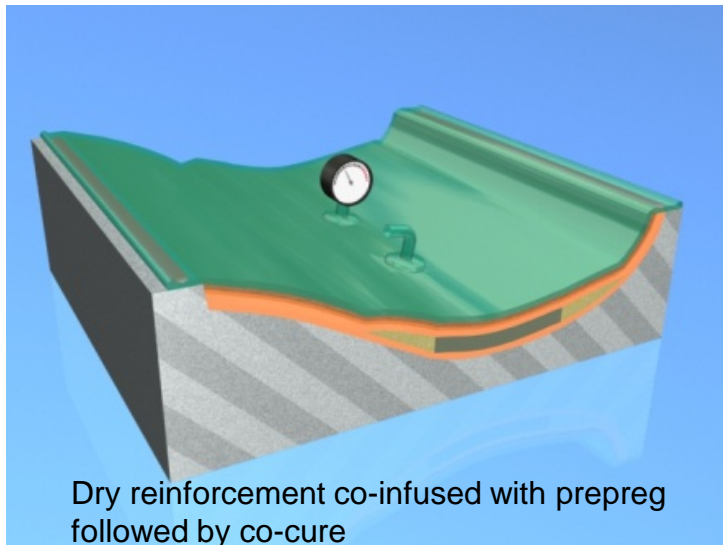
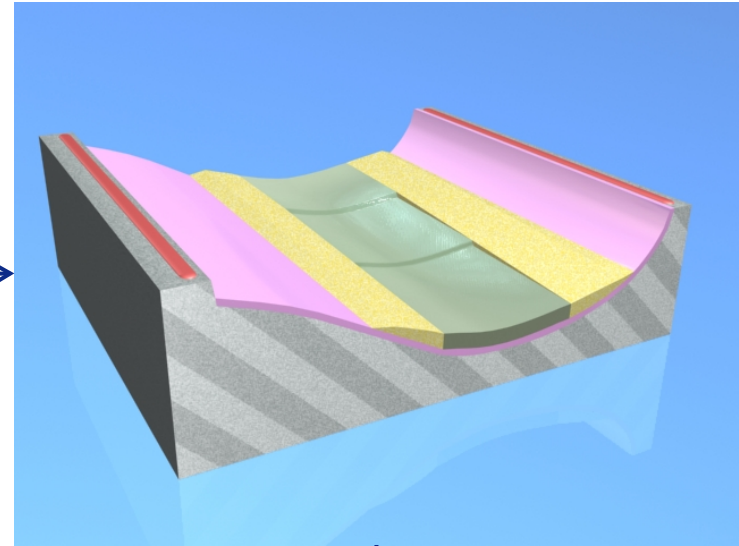
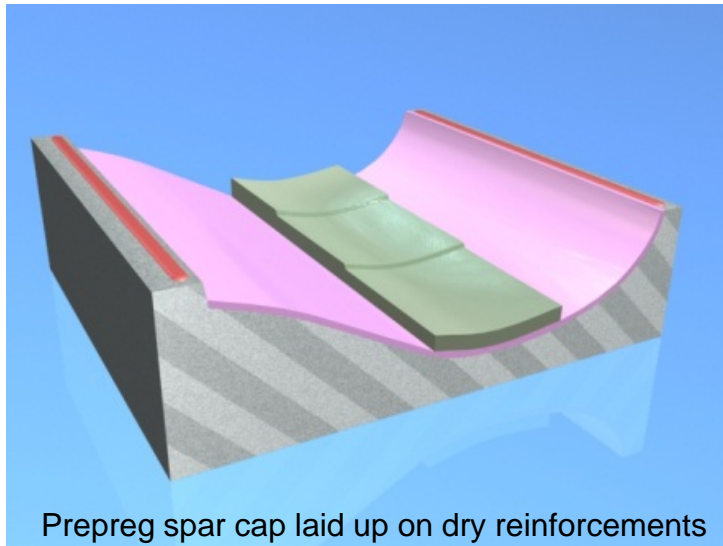
UD prepreg for the heavy load-carrying structure

Infusion of dry reinforcement for the remainder of the structure

Cure of the whole assembly at the same time and temperature

M79 simplifies co-infusion when making large structures because it cures at 70-80°C (i.e. same temperature for both infusion and prepreg matrices)

Wind Blades: M79 co-cured in an Infused Shell



Co-infusion: Case Study after Demoulding

The finished 4x2m laminate



Low porosity, high Tg

FV (%)		50
Porosity (%)	Side	0,7
	Middle	1,5
Tg (°C)	Top	75
	Middle	120
	Bottom	75
Cure cycle		6hrs 90°C

Co-infusion simplifies the production process, combining the strengths of prepreg and infusion materials

Conclusions

- **Prepregs are used for both structural and surface applications in wind blade construction**
- **The surface of XF2P laminates have been characterised for surface defects**
 - Defects can be reduced by >99% to give a paint ready surface
- **M79, a new matrix for wind blades, has been characterised after cure at 70° and 80°C**
 - Cure reaction enthalpies are 100-120 j/g, reducing from 350 j/g over the last 15 years
 - Static mechanical properties compare favourably with standard materials
- **The low cure temperature of M79 helps enable co-infusion of prepreg with dry reinforcements, thus combining the best features of each process**

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